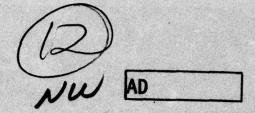
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**MEMORANDUM REPORT NO. 2711** 

NEWT: A COMPUTER CODE FOR CALCULATING THE KINEMATIC VARIABLES OF A PROJECTILE SUBJECT TO A SECOND ORDER DRAG LAW THAT IS PENETRATING AN ELASTIC-PLASTIC MEDIUM

William B. Beverly

December 1976

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USA BALLISTIC RESEARCH LABORATORIES ABERDEEN PROVING GROUND, MARYLAND



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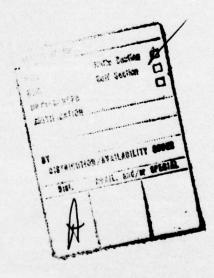
MONITORING AGENCY NAME & ADDRESS(If different from Controlling Office) 15. SECURITY CLASS. (of this report) UNCLASSIFIED DECLASSIFICATION/DOWNGRADING 16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited. -MR-2711 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Ballistic Penetration Elastic-Plastic Medium Differential Equation Second Order Drag Force ABSTRACT (Continue on reverse side H necessary and identity by block number) The motion of a projectile in an elastic-plastic medium can often be adequately described by a second order drag law. The differential equations describing this motion have been solved for a projectile of impacting velocity Vo and frontal area R and a computer program NEWT has been written in the BASIC language to calculate the penetration, velocity, and energy deposition of the projectile at subsequent times. The code was written to describe projectiles in tissue simulant but it can be used to predict any motion described by a econd-order drag UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Date !

UNCLASSIFIED SECURITY/CLASSIFICATION OF THIS PAGE(When Date Entered) The differential equations and their solutions are discussed in detail. A flow chart and code listing are given. A description of the user-supplied input is given. Representative test problems are solved and their solutions are discussed.

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### I. INTRODUCTION

A general empirical expression for the motion of a projectile of mass m and velocity V obeying a second order drag law has been proposed that has the form  $^{\rm l}$ 

$$F = m\frac{dV}{dT} = -A^{\dagger}V^{2} - B^{\dagger}V - C^{\dagger}.$$
 (1)

The coefficients A', B', and C' are quantities that can be either evaluated experimentally from penetration-time data or calculated using basic quantities. The terms on the right side of Eq. 1 can be considered to be the forces contributing to three components of energy deposition. These three components or modes of energy deposition are: (1) the acceleration of target material adjacent to the projectile, (2) the effect of frictional forces, and (3) the cohesive strength of the target material respectively. A comprehensive bibliography of earlier work using versions of Eq. 1 where one or two of the coefficients were set to zero is given by Goldsmith<sup>2</sup> on p.298.

Dienes and Welch<sup>3</sup> define the coefficients in terms of basic physical quantities. Their definitions are:

$$A' = \frac{\pi}{2} \rho C_D r^2 \tag{1A}$$

$$B' = 6\pi r \eta \tag{1B}$$

$$C' = \pi r^2 \sigma \tag{1C}$$

where  $\rho$  is the target density and  $\pi r^2$  is the frontal area of the projectile. The quantity  $C_D$  is the drag coefficient,  $\eta$  is the dynamic viscosity, and  $\sigma$  is the normal stress.

The rate of work by the projectile as it penetrates the target is equal to its rate of loss of kinetic energy and is given by

$$F \cdot dS = d(\frac{1}{2} mV^2) \tag{2A}$$

<sup>1.</sup> W. A. Allen, E.B. Mayfield, and H.L. Morrison, "Dynamics of a Projectile Penetrating Sand," Journal of Applied Physics, 28, 1957.

<sup>2.</sup> Weiner Goldsmith, "Impact," Edward Arnold (Publishers) Ltd., London, England, 1960.

<sup>3.</sup> J.K. Dienes and J.E. Welch, "Parameter Study of Rigid Body Penetration," Systems, Science and Software Report No. SSS-R-75-2521, 1974.

This leads in the limit to

$$F = mV \frac{dV}{dS} = -A'V^2 - B'V - C'$$
 (2B)

Eqs. 1 and 2B are solved for an impacting velocity  $\mathbf{V}_{\mathbf{O}}$  at time zero to obtain analytical relations for the velocity and penetration as functions of time. The solutions are discussed in the next section.

A computer code NEWT was written, using the BASIC programming language, to calculate the penetration and velocity values for a given impacting velocity and a given set of coefficients at times subsequent to impact. The energy deposited (both total and the components contributed by each force component) along the projectile trajectory are also calculated.

A listing of the code is given in Appendix B and a description of the code that includes a flow chart is given in Section IV. A description of the code input, to be supplied by the user, is given in Section V. Sample problems that demonstrate the differing effect of each coefficient on the projectile penetration are discussed in Appendix A.

The code has been run successfully on the WANG 2200 calculator. Results and identifying captions are printed by the calculator-controlled typewriter. Approximately 5000 words of memory are needed to execute the program.

### II. CALCULATIONAL PROCEDURE

Equation 1 may be expressed in integral form as

$$\int_{V_{O}}^{V} \frac{dV}{AV^{2}} \frac{dV}{BV + C} = - \int_{O}^{T} dT = -T .$$
 (3)

where  $V_{\rm O}$  and V are the velocities at time zero and T respectively. The constant m has been absorbed into the coefficients yielding the unprimed set. The integrals can be solved in closed form where the solution to the leftmost integral may assume one of the three forms dependent upon the value of the discriminator RI where

$$R1 = B^2 - 4AC . (4)$$

The three forms are:

1. R1 > 0,  

$$T = \left| \frac{1}{R} \ln \frac{2AV + B - R}{2AV + B + R} \right|_{V}^{V_{O}},$$
(5A)

2. R1 < 0,  

$$T = \left| \frac{2}{R} \tan^{-1} \frac{2AV + B}{R} \right|_{V}^{V_0}$$
, (5B)

3. R1 = 0,  

$$T = \left| \frac{-2}{2AV + B} \right|_{V}^{V}$$
 (5C)

The quantity R is the positive real value of

$$R = \sqrt{\pm R1} \quad . \tag{6}$$

Eq. 2B in integral form can be expressed as

$$\int_{V}^{V_{0}} \frac{V dV}{AV^{2} + BV + C} = -\int_{S}^{0} dS = S .$$
 (7)

The solution to this integral is

$$S = \left| \frac{1}{2A} \ln(AV^2 + BV + C) \right|_{V}^{V_0} - \frac{B}{2A} \int_{V}^{V_0} \frac{dV}{AV^2 + BV + C} . \tag{8}$$

It should be noted that the rightmost integral has the same form as that derived in Eq. 3. This permits Eq. 8 to be expressed as

$$S = \left| \frac{1}{2A} \ln (AV^2 + BV + C) \right|^{0} - \frac{BT}{2A}$$
 (9)

or

$$S = \frac{1}{2A} \left[ an \frac{AV_0^2 + BV_0 + C}{AV^2 + BV + C} - BT \right] . \tag{10}$$

Returning to Eq. 3, its solution for positive values of R1 is given by

$$T = \left| \frac{1}{R} \ln \frac{2AV + B - R}{2AV + B + R} \right|^{V_{0}}$$
 (11)

which expands to

$$T = \frac{1}{R} \ln \left( \frac{1}{G} \frac{2AV + B + R}{2AV + B - R} \right)$$
 (12A)

where

$$G = \frac{2AV_{o} + B + R}{2AV_{o} + B - R} . {12B}$$

The term G will always be positive. The velocity is given explicitly by

$$V = \frac{R + B + Ge^{RT}(R - B)}{2A(Ge^{RT} - 1)} . (13)$$

That relation, as expected, will give a value of  $V_{\rm o}$  to the velocity when T is zero.

The solution to Eq. 3 for negative values of Rl is given by

$$T = \left| \frac{2}{R} \tan^{-1} \frac{2AV + B}{R} \right|_{V}^{V_{O}}$$
 (14)

which expands to

$$T = \frac{2}{R} \left( \tan^{-1} \frac{2AV_0 + B}{R} - \tan^{-1} \frac{2AV + B}{R} \right) .$$
 (15)

The velocity is given explicitly by

$$V = \frac{R}{2A} \left[ \tan(\tan^{-1} \frac{2AV_0 + B}{R} - \frac{RT}{2}) - \frac{B}{R} \right] .$$
 (16)

The solution to Eq. 3 for a zero value of R1 is given by

$$T = \left| \frac{-2}{2AV + B} \right|_{V}^{V_0}$$
 (17)

which expands to

$$T = 2 \left[ \frac{1}{2AV + B} - \frac{1}{2AV_0 + B} \right] . {18}$$

The velocity is given explicitly by

$$V = \frac{4AV_0 - BT(2AV_0 + B)}{2A[2 + T(2AV_0 + B)]} . (19)$$

The preceding solutions are valid only for positive, non-zero values of the coefficient A. The integral in Eq. 3 was solved for zero values of A and positive, non-zero values of B yielding

$$V = \frac{1}{B} \left[ e^{-BT} (BV_0 + C) - C \right] . (20)$$

Similarly, Eq. 8 was solved for the penetration yielding

$$S = \frac{1}{B} \left[ V_{o} - V + \frac{C}{B} \ln \frac{BV + C}{BV_{o} + C} \right] . \qquad (21)$$

None of the preceding solutions are valid if both A and B are zero. For this case the solution to Eq. 3 was found to be

$$V = V_0 - CT (22)$$

The solution for the penetration was

$$S = \frac{1}{2C} \left[ V_0^2 - V^2 \right] . {(23)}$$

### III. CODE DESCRIPTION

The code NEWT was designed so that it would check the input values of A, B, and C, evaluate the discriminator if necessary, choose the proper set of equations to use, and calculate the penetration, velocity, and energy deposition rates at the chosen times. The input and the calculated results are then printed in tabular form along with their captions.

A flow chart of NEWT is displayed in Figure 1. The important variables and arrays are listed and briefly described in Table I.

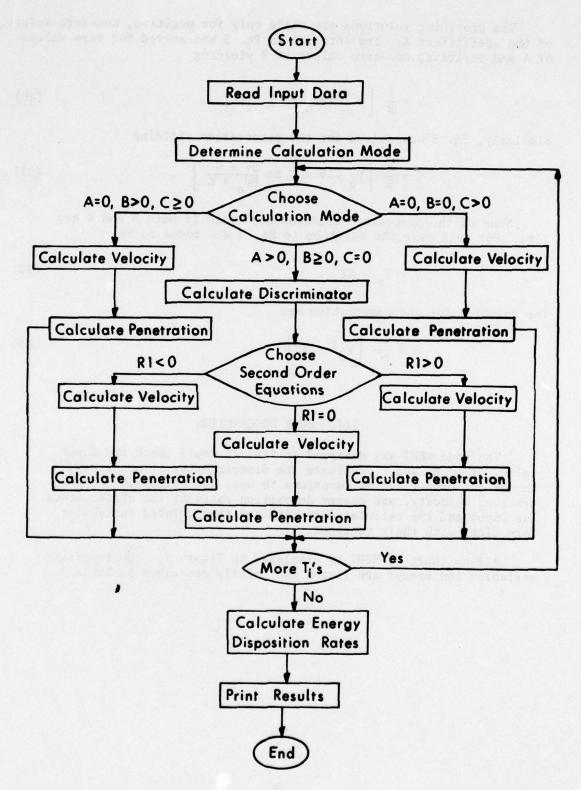


Figure 1. NEWT Flow Chart

Table I. Variables and Arrays

ariable	Description
A3	The coefficient A in Equation 1
В3	The coefficient B in Equation 1
23	The coefficient C in Equation 1
4 -3-11-330	$A = \overline{A3} \text{ M/0.5} \pi R^2$
3	$B = \overline{B3} M/6\pi R$
Thron Care	$C = C3 M/\pi R^2$
<b>(</b>	Effective impacting radius of Projectile
M (S)	Mass of Projectile
0	Initial Velocity of Projectile
7	Calculation mode flag
	Q7 = 001, A = B = 0, C > 0
	Q7 = 010, A = C = 0, B > 0
	Q7 = 011, A = C = 0, C > 0
	Q7 = 100, A > 0, B = C = 0
	Q7 = 101, A > 0, B = 0, C > 0
	Q7 = 111, A > 0, B > 0, C > 0
1 percent of teach	. Calculation discriminator
	$F1 = 1$ , $B^2 - 4AC > 0$
	$F1 = 2$ , $B^2 - 4AC < 0$
	$F1 = 3$ , $B^2 - 4AC = 0$
11	Number of times
	Times
r1 (50)	Times array
I	Velocity
/1 (50)	Velocity array
S	Penetration
S1 (50)	Penetration array
R1	$R1 = \sqrt{\pm (B^2 - 4AC)}$

### IV. INPUT

The user-supplied input needed to execute the program is listed in Table II. It is listed in the order by which it is requested by NEWT. CGS units should be used to conform to captions but they could be easily changed to conform to another system. It should be noted that the programming variables A, B, and C are not the coefficients A, B, and C. The coefficients A, B, and C correspond to the programming variables A3, B3, and C3.

Table II. Code-Generated Queries

	Instruction	Description
1.	Title =	Problem Title
2.	A =	$A = \rho C_D  (Eq. 1A)$
3.	B = 91.700fox7 10 a	$B = \eta(Eq. 1B)$
4.	C = Manageri de Valoniel lara	$C = \sigma(Eq. 1C)$
5.	Radius =	Frontal Area = $\pi r^2$
6.	Mass =	Mass = (gm)
7.	Initial Velocity =	$V_0 = (cm/sec)$
8.	No Times =	No of times for which kinetic
		quantities are to be calculated
9.	1 for Read-in Times =	F2 = 0 Calculates times
10.	Times Interval =	Time Interval (calculated times only)
11.	Next Time =	Next Time (Read-in Times only)

### V. CONCLUSIONS

This study, and the building of the Computer Code, were conducted at the Target Assessments Branch (TAB), Vulnerability/Lethality Division(VLD), Ballistic Research Laboratories (BRL) as part of an on-going effort to calculate the penetration of human tissue by bullets or fragments and to derive the resulting incapacitation probabilities. However, the code is not restricted to a tissue media, but can be used to calculate the penetration of any media that is described by a second order drag law. A magnetic tape copy of the code can be obtained by sending a blank computer tape cassette to the TAB.

Earlier work by Bruchey 4, in performing least square determination of drag coefficients for handgum bullets penetrating tissue simulant, has provided a validation of the model for non-zero values of the A-and C-coefficients and a zero value of the B-coefficients. It would therefore be expected that a better fit of experimental data to the model, as well as a better description of the energy deposition modes in the medium, could be obtained by optimizing all three coefficients instead of only the A-coefficients as was done in the preceding work. A computer code for providing this least squares fit of experimentally-determined penetration-time data to the model by optimizing all three coefficients is being developed by the TAB. 5

The degree of human incapacitation (however this term is ultimately quantized and defined) will probably be the sum of the products formed by the volume of a particular tissue destroyed and its vulnerability index. The volume of tissue destroyed is approximated by the volume of the maximum temporary wound cavity (TWC). A future effort will be made to construct a wound cavity model that used the energy deposition rates to predict the size of the TWC. It is expected that the forthcoming study and the resulting Computer Code will provide a rapid and inexpensive means of calculating the volume of that cavity.

William J. Bruchey, Jr., "Effective Drag Coefficients for Various Small Arms Projectiles in Dense Media," BRL Memorandum Report 2481, May 1975.

(AD #B004154L)

<sup>&</sup>lt;sup>5</sup> William B. Beverly, "LSP: A Least Square Program for Calculating The Resistive Force Coefficients of a Projectile Subject to a Second Order Drag Law," to be published as a BRL Report.

### ACKNOWLEDGEMENT

The author wishes to acknowledge the contribution of Mr. Eugene T. Roecker whose suggestion greatly simplified the mathematical equations and permitted more efficient computer coding of NEWT.

APPENDIX A
SAMPLE PROBLEMS

### APPENDIX A

### SAMPLE PROBLEMS

Sample values A, B, and C were chosen so that the three components of the opposing force would be equal at the impacting velocity of 30,000 cm/sec. This choice of input values permits an easy analysis of the kinematic results for demonstrating the differing effects of the three force components. The penetration and velocity versus time are listed in Table A-I, the partial differential (in space) energy losses are listed in Table A-II, and the partial cumulative energy losses are listed in Table A-III.

Three more calculations were conducted in which A, B, and C in turn were assigned a non-zero value (that used in the first problem) while the remaining two coefficients were set to zero. The penetration and velocity versus time values calculated for these problems are listed in Tables A-IV - A-VI. The penetration and velocity versus time for all test problems are displayed in Figure A-1.

Table A-I. Kinematic Results (A>0, B>0, C>0)

MASS= 8.0	OOOE+OO RADIUS=	6.000E-01 INIT	VEL= 3.000E+04
A= 1.2	200E+00 B=	1.800E+03 C=	5.399E+03
A3= 8.4	82E-02 E3=	2.544E+03 C3=	7.633E+07
KINEMATIC RES	SULTS		
TIME (SEC)	PEN(CM)	VEL (CH/SEC)	DEL E/DEL S
0.0000E+00	0.0000E+00	3.0000E+04	2,2901E+03
1.0000E-05	2.3883E-01	2.7793E+04	2.1258E+0S
2.0000E-05	5.5638E-01	2.5741E+04	1.9005E+03
3.0000E-05	8.0412E-01	2.3827E+04	1.3512E+08
4.0000E-05	1.0333E+00	2.2034E+04	1.7359E+03
5.0000E-05	1.2451E+00	2.0351E+04	1.6325E+00
6.0000E-05	1.4406E+00	1.3766E+04	1.5396E+03
7.00001-05	1.6208E+00	1.7269E+04	1.4557E+08
8.00001-05	1.7863E+00	1.5851E+04	1.3799E+03
9.0000E-05	1.93801:+00	1.4506E+04	1.3110E+02
1.0000E-04	2.0767E+00	1.3227E+04	1.2484E+03
1.1000E-04	2.2028E+00	1.2008E+04	1.1912E+08
1.2000E-04	2.31.70E+00	1.0843E+04	1.1390E+08
1.3000E-04	2.4198E+00	9.7287E+03	1.0012E+03
1.4000E-04	2.5117E+00	8.6597E+03	1.0473E+00
1.5000E-04	2.5932E+00	7.632SE+03	1.0070E+03
1.6000E-04	2.6645E+00	6.6446E+03	9.6991E+07
1.7000E-04	2.7262E+00	5.6920E+03	9.3570E+07
1.3000E-04.	2.7785E+00	4.7723E+03	9.0413E+07
1.9000E-04	2.82171:+00	3.882911+03	8.7497E+07
2.0000E-04	2.8562E+00	3.0216E+03	8.4801E+07
2.1000E-04	2.882211+00	2.1862E+03	8.2306E+07
2.2000E-04	2.9000E+00	1.3749E+03	7.0006E+07
2.3000E-04	2.9093E+00	5.8577E+02	7.7857E+07

Table A-II. Differential Energy Deposition (A>0, B>0, C>0)

TIME (SEC)	A3*V*V	B3*V	C3
0.0000E+00	7.6340E+07	7.6340E+07	7.6337E+07
1.0000E-05	6.5524E+07	7.0726E+07	7.6337E+07
2.0000E-05	5.6207E+07	6.5505E+07	7.6337E+07
3.0000E-05	4.8157E+07	6.0633E+07	7.6337E+07
4.0000E-05	4.1184E+07	5.6071E+07	7.6337E+07
5.0000E-05	3.5131E+07	5.1788E+07	7.6337E+07
6.0000E-05	2.9871E+07	4.7753E+07	7.6337E+07
7.0000E-05	2.5296E+07	4.3944E+07	7.6337E+07
8.0000E-05	2.1314E+07	4.0338E+07	7.6337E+07
9.0000E-05	1.7851E+07	3.6915E+07	7.6337E+07
1.0000E-04	1.4841E+07	3.3660E+07	7.6337E+07
1.1000E-04	1.2231E+07	3.0557E+07	7.6337E+07
1.2000E-04	9.9737E+06	2.7593E+07	7.6337E+07
1.3000E-04	8.0283E+06	2.4756E+07	7.6337E+07
1.4000E-04	6.3610E+06	2.2036E+07	7.6337E+07
1.5000E-04	4.9418E+06	1.9423E+07	7.6337E+07
1.6000E-04	3.7450E+06	1.6908E+07	7.6337E+07
1.7000E-04	2.7482E+06	1.4484E+07	7.6337E+07
1.8000E-04	1.9318E+06	1.2144E+07	7.6337E+07
1.9000E-04	1.2789E+06	9.8809E+06	7.6337E+07
2.0000E-04	7.7447E+05	7.6891E+06	7.6337E+07
2.1000E-04	4.0544E+05	5.5634E+06	7.6337E+07
2.2000E-04	1.6034E+05	3.4987E+06	7.6337E+07
2.3000E-04	2.9105E+04	1.4906E+06	7.6337E+07

Table A-III. Cumulative Energy Deposition (A>0, B>0, C>0)

CUMULATIVE TIME (SEC)	ENERGY LOSS (ERGS) INT (A3*V*V) DS	INT(B3*V)DS	INT (C3) DS
0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00
1.0000E-05	1.6390E+08	1.6991E+08	1.7639E+08
2.0000E-05	2.9418E+08	3.1570E+08	3.3978E+08
3.0000E-05	3.9760E+08	4.4070E+08	4.9108E+03
4.0000E-05	4.7951E+08	5.4770E+08	6.3106E+08
5.0000E-05	5.4418E+08	6.3910E+08	7.6043E+08
6.0000E-05	5.9502E+08	7.1695E+08	8.7983E+08
7.0000E-05	6.3476E+08	7.8301E+08	9.8982E+08
8.0000E-05	6.6562E+08	8.3882E+08	1.0909E+09
9.0000E-05	6.8940E+08	8.8571E+08	1.1835E+09
1.0000E-04	.7.0752E+08	9.2484E+08	1.2682E+09
1.1000E-04	7.2118E+08	9.5724E+08	1.3452E+09
1.2000E-04	7.3133E+08	9.8381E+08	1.4150E+09
1.3000E-04	7.3873E+08	1.0053E+09	1.4778E+09
1.4000E-04	7.4402E+08	1.0225E+09	1.5339E+09
1.5000E-04	7.4770E+08	1.0360E+09	1.5836E+09
1.6000E-04	7.5018E+08	1.0464E+09	1.6272E+09
1.7000E-04	7.5178E+08	1.0541E+09	1.6649E+09
1.8000E-04	7.5276E+08	1.0597E+09	1.6968E+09
1.9000E-04	7.5332E+08	1.0635E+09	1.7232E+09
2.0000E-04	7.5360E+08	1.0659E+09	1.7443E+09
2.1000E-04	7.5372E+08	1.0673E+09	1.7602E+09
2.2000E-04	7.5376E+08	1.0679E+09	1.7710E+09
2.3000E-04	7.5377E+08	1.0681E+09	1.7770E+09

Table A-IV. Kinematic Results (A>0, B=0, C=0)

TEST PROBLE MASS=	EM 2 8.000E+00 RADIUS=	6.000E-01 TUTT	VEL= 3.000E+04
	1.200E+00 B=	0.000E+00 C=	0.000E+00
	8.482E-02 B3=	0.000E+00 C3=	0.000E+00
KINEMATIC:	AND THE RESERVE OF THE PARTY OF		
TIME (SEC)	PEN(CH)	VEL (CM/SEC)	DEL E/DEL S
0.0000E+00	0.0000E+00	3.0000E+04	7.6340E+07
1.0000E-05	2.9624E-01	2.9255E+04	7.2598E+07
2.0000E-05	5.8523E-01	2.8547E+04	6.9125E+07
3.0000E-05	8.6730E-01	2.7872E+04	6.5895E+07
4.0000E-05	1.1427E+00	2.7228E+04	6.2886E+07
5.0000E-05	1.4119E+00	2.6613E+04	6.0079E+07
6.0000E-05	1.6751E+00	2.6026E+04	5.7456E+07
7.0000E-05	1.9325E+00	2.5464E+04	5.5001E+07
8.0000E-05	2.1845E+00	2.4925E+04	5.2699E+07
9.0000E-05	2.4311E+00	2.4409E+04	5.0540E+07
1.0000E-04	2.6727E+00	2.3914E+04	4.8510E+07
1.1000E-04	2.9095E+00	2.3439E+04	4.6600E+07
1.2000E-04	3.1416E+00	2.2982E+04	4.4801E+07
1.3000E-04	3.3692E+00	2.2542E+04	4.3104E+07
1.4000E-04	3.5925E+00	2.2119E+04	4:1502E+07
1.5000E-04	3.8116E+00	2.1712E+04	3.9987E+07
1.6000E-04	4.0268E+00	2.1319E+04	3.8554E+07
1.7000E-04	4.2381E+00	2.0940E+04	3.7196E+07
1.8000E-04	4.4456E+00	2.0575E+04	3.5909E+07
1.9000E-04	4.6496E+00	2.0222E+04	3.4688E+07
2.0000E-04	4.8501E+00	1.9881E+04	3.3528E+07
2.1000E-04	5.0473E+00	1.9551E+04	3.2425E+07
2.2000E-04	5.2412E+00	1.9232E+04	3.1376E+07
2.3000E-04	5.4320E+00	1.8924E+04	3.0377E+07

Table A-V. Kinematic Results (B>0, A=C=0)

	8.000E+00 RADIUS=		VEL= 3.000E+04
SERVICE MENTS VIEW, CARROLL MINES		1.800E+03 C=	0.000E+00
A.3=	0.000E+00 B3=	2.544E+03 C3=	0.000E+00
KINEMATIC	RESULTS		
TIME (SEC)	PEH (CH)	VEL (CM/SEC)	DEL E/DEL S
0.0000E+00	0.0000E+00	3.0000E+04	7.6340E+07
1.0000E-05		2.9246E+04	7.4422E+07
2.0000E-05		2.8511E+04	7.2552E+07
3.0000E-05		2.7795E+04	7.0729E+07
4.0000E-05		2.7096E+04	6.8952E+07
5.0000E-05		2.6415E+04	6.7220E+07
6.0000E-05		2.5752E+04	6.5531E+07
7.0000E-05		2.5105E+04	6.3884E+07
3.0000E-05	2.1714E+00	2.4474E+04	6.2279E+07
.0000E-05	2.4131E+00	2.3859E+04	6.0714E+07
1.0000E-04	2.6487E+00	2.3259E+04	5.9139E+07
1.1000E-04	2.8783E+00	2.2675E+04	5.7701E+07
L.2000E-04	3.1022E+00	2.2105E+04	5.6252E+07
L.3000E-04	3.3205E+00	2.1550E+04	5.4838E+07
L.4000E-04	3.5333E+00	2.1008E+04	5.3460E+07
L.5000E-04	3.7407E+00	2.0480E+04	5.2117E+07
L.6000E-04		1.9966E+04	5.0808E+07
L.7000E-04		1.9464E+04	4.9531E+07
L.3000E-04		1.8975E+04	4.8286E+07
1.9000E-04		1.8498E+04	4.7073E+07
2.0000E-04		1.8034E+04	4.5890E+07
2.1000E-04		1.7580E+04	4.4737E+07
2.2000E-04		1.7139E+04	4.3613E+07
2.3000E-04	5.2232E+00	1.6708E+04	4.2518E+07

Table A-VI. Kinematic Results (C>0, A=B=0)

TEST PROBL			0 = 1.(2)
	8.000E+00 RADIUS=		VEL= 3.000E+04
	0.000E+00 E=	0.000E+00 C=	5.399E+03
A3=	0.000E+00 B3=	0.000E+00 C3=	7.633E+07
KINEMATIC	RESULTS		
TIME (SEC)	PEH (CH)	VIL (CM/SEC)	DEL E/DEL S
0.0000E+00	0.0000E+00	3.0000E+04	7.6337E+07
1.0000E-05	2.9613E-01	2.9236E+04	7.6337E+07
2.0000E-05		2.84731:+04	7.6337E+07
3.0000E-05	8.6564E-01	2.7709E+04	7.6337E+07
4.0000E-05	1.1389E+00	2.6946E+04	7.63371:+07
5.0000E-05	1.4045E+00	2.6183E+04	7.6337E+07
6.0000E-05	1.6625E+00	2.5419E+04	7.6337E+07
7.0000E-05	1.9129E+00	2.4656E+04	7.6337E+07
8.0000E-05	2.1557E+00	2.3892E+04	7.6337E+07
9.0000E-05	2.3908E+00	2.3129E+04	7.6337E+07
1.0000E-04	2.6183E+00	2.2366E+04	7.6337E+07
1.1000E-04	2.8381E+00	2.1602E+04	7.6337E+07
1.2000E-04	3.05031:+00	2.0839E+04	7.6337E+07
1.3000E-04	3.2549E+00	2.0076E+04	7.6337E+07
1.4000E-04	3.4518E+00	1.9312E+04	7.6337E+07
1.5000E-04		1.8549E+04	7.6337E+07
1.6000E-04		1.7785E+04	7.6337E+07
1.7000E-04		1.7022E+04	7.6337E+07
1.8000E-04		1.6259E+04	7.6337E+07
1.9000E-04		1.5495E+04	7.6337E+07
2.0000E-04		1.4732E+04	7.6337E+07
2.1000E-04		1.3969E+04	7.6337E+07
2.2000E-04		1.3205E+04	7.6337E±07
2.3000E-04	4.8808E+00	1.2442E+04	7.6337E+07

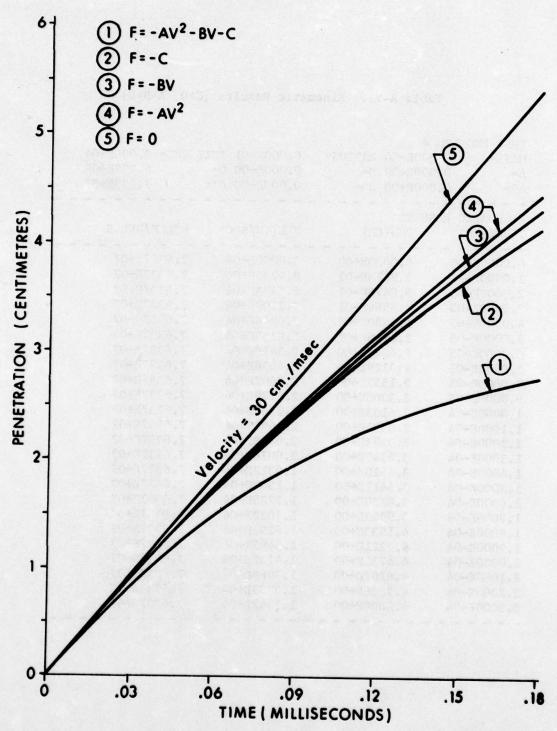


Figure A-1. Results of Test Problems

APPENDIX B
NEWT PROGRAM LISTING

## APPENDIX B

### NEWT PROGRAM LISTING

TREM NEWT3 2REM F=-A\*V\*V-B\*V-C 31NPUT "TITLE",Z2\$,Z3\$,Z4\$,Z5\$ 4SELECT LIST 211 **5SELECT PRINT 211 GSELECT R** 10DIN T1(50),V1(50),S1(50)
15INPUT "A=",A
20INPUT "B=",B 25INPUT "C=",C
30INPUT "RADIUS=",R
35INPUT "MASS=",M
40INPUT "INITIAL VELOCITY=",VO 45 INPUT "NO. TIMES=",N1 50INPUT "1 FOR READ-IN TIMES",F2 55IF F2=0THEN 75 60FOR I=1TO N1 65 INPUT "NEXT TIME=",T1(I) 70NEXT I 75A3=#PI\*0.5\*R\*R\*A/M 80B3=6.0\*#PI\*R\*B/M 85C3=#PI\*R\*R\*C/M 901F F2]OTHEN 125 95T=0.0 100IMPUT "TIME INTERVAL=", Z1 105FOR I=1TO 111 110T1(I)=T 115T=T+Z1 120NEXT I 125 IF A]OTHEN 134 1271F B]OTHEN 130 12807=001: GOTO 220 130IF C]OTHEN 132 13107=010: GOTO 220 13207=011: GOTO 220 1341F B]OTHEN 138 1351F C]OTHEN 137 13607=100: GOTO 154 13707=101: GOTO 154 1381F C]OTHEN 140

```
13907=110: GOTO 154
14007=111
154REM CALCULATE B*B-4*A*C FLAG
155F1=B3*B3-4.0*A3*C3
1601F F1[0.0THEN 175
1651F F1 0.0THEN 130
170F1=3:GOTO 220
175F1=2:GOTO 220
180F1=1
220REIL PRINT INPUT AND CAPTIONS
222PRINT Z25,Z35,Z45,Z5$
223Z2$="MASS="
224Z3$="RADIUS="
225Z4$="INIT VEL="
226PRINTUSING 227,Z20,M,Z35,R,Z45,V0
227844444444 4.44411!! ******* 4.***** 4.***** 4.***** 4.***!!!!
228Z2$="A="
229Z3$="B="
23074$="C="
231PRINTUSING 227, Z2$, A, Z3$, B, Z4$, C
232Z2$="A3="
233Z3S="B3="
234Z4$="C3="
235PRINTUSING 227, Z2$, A3, Z3$, B3, Z4$, C3
245GOSUR '01
246PRINT "KINEMATIC RESULTS"
250Z1S="TIME(SEC)":Z2$="PEH(CM)":Z3$="VEL(CM/SEC)"
251Z4S="DEL E/DEL S"
255PRINTUSING 260,Z1$,Z2$,Z3$,Z4$
260%##########
                   444444
                                    4444444444
                                                    ##########
265GOSUB '01
270FOR I=1TO [1]
2751F Q7]1THEN 295
280GOSUB '11(T1(I))
285GOSUD '21
290G0T0 340
2951F Q7]11THEN 315
300GOSUB '13(T1(I))
305GOSUB '23
310G0T0 340
315IF Q7]111THEN 335
320GOSUB '16(T1(I))
325GOSUB '26
330G0T0 340
335A9=2:GOTO 430
340NEXT I
342IF T1(1)]0.0THEN 345
343S1(1)=0.0
345FOR I=1TO 111
346IF V1(I)[OTHEN 352
```

```
350PRINTUSING 355,T1(I),S1(I),V1(I),A3*V1(I)!2+B3*V1(I)+C3
351G0T0 360
352PRINT "NEGATIVE VELOCITY"
353N1=I-1
354G0T0 365
355%#.####!!!! #.####!!!! #.####!!!! #.####!!!!
360NEXT I
365GOSUB '01
367FOR I=1TO 5
368PRINT
369NEXT I
370GOSUB '01
371Z2$="A3*V*V": Z3$="B3*V": Z4$="C3"
372PRINT "DIFFERENTIAL ENERGY LOSS(ERGS/CM)"
373PRINTUSING 374,Z1$,Z2$,Z3$,Z4$
373PRINTUSING 374,Z1$,Z2$,Z3$,Z4$
                                 #### ##
374%##########
                   ######
375GOSUB '01
376FOR I=1TO N1
378PRINTUSING 355,T1(I),A3*V1(I)*V1(I),B3*V1(I),C3
380NEXT I
382GOSUB '01
384FOR I=1TO 5
386PRINT
383NEXT I
389GOSUB '01
390PRINT "CUMULATIVE ENERGY LOSS(ERGS)"
391Z2S="INT(A3*V*V)DS"
392Z3$="INT(B3*V)D$"
393Z4$="INT(C3)DS"
396PRINTUSING 398, Z1$, Z2$, Z3$, Z4$
                   398%###########
399GOSUB '01
400Z6=T1(1)
402Z7=0.5*A3*(V0!2+V1(1)!2)*S1(1)*M
404Z8=0.5*B3*(V0+V1(1))*S1(1)*M
406Z9=C3*S1(1)*M
408PRINTUSING 355, Z6, Z7, Z8, Z9
410FOR I=2TO N1
412Z6=T1(I)
414Z7=Z7+0.5*A3*(V1(I-1)!2+V1(I)!2)*(S1(I)-S1(I-1))*M
416Z8=Z8+0.5*B3*(V1(I-1)+V1(I))*(S1(I)-S1(I-1))*M
413Z9=Z9+C3*(S1(I)-S1(I-1))*M
420PRINTUSING 355,Z6,Z7,Z8,Z9
422NEXT I
424GOSUB '01
430END
5100DEFFN'11(T)
5102REM CALCULATES VELOCITY Q7=001
5105V=V0-C3*T
5110V1(I)=V
```

```
5115RETURN 5300DEFFN'13(T)
5302REM CALCULATES VELOCITY 07=011
5305V=B3*T:V=EXP(-V)

5310V=V*(B3*V0+C3)

5315V=(V-C3)/B3

5320V1(I)-V
5320V1(I)=V
5325RETURN
56UODEFFN'16(T)
5600DEFFN'16(T)
5602REN CALCULATES VELOCITY Q7=111
5605IF F1]1THEN 5645
5607REM LOG FURM

5610R1=B3!2-4.0*A3*C3:R1=SQR(R1)

5615F3=2.0*A3*V0+B3+R1

5620F3=F3/(F3-2.0*R1)

5625V=EXP(R1*T): V=V*F3*(R1-B3)

5630V=V+B3+R1

5635 V=V/(2.0*A3*(F3*EXP(R1*T)-1.0))
5640G0T0 5694
5645IF F1]2THEN 5680
5647BF11 TRIG FORM
5647REH TRIG FORM
5650R1=4.0*A3*C3-B3!2: R1=SQR(R1)
5655V=2.0*A3*V0+B3:V=V/R1
5660V=ACCTAN(V): V=V-R1*T/2.0
5665V=TAN(V)-B3/R1
5665V=TAN(V)-B3/R1
5670V=0.5*R1*V/A3
5670V=0.5*R1*V/A3
5675G0T0 5694
5680IF F1]3THEN 5692
5682REM ALGEBRAIC FORM
5685V=2*A3*V0+B3
5687V=2*A3*(2+T*V)
5690V=(4*A3*V0+B3*T*(2*A3*V0+B3))/V
5692A9=1*G0T0_b696
5692A9=1:00T0 5696
                ## (1) 12* | $1(1) 14+$10V) #EA#8, 0+1 1000

## (1) 12* ((1) 14+$V) #EA#8, 0+1 1000
5694V1(I)=V
509GRETURN
6100 DEFFIL'21
G102HE!: CALCULATES PENETRATIO!! Q7=001
6105S=V0*V0-V*V
6110S=0.5*S/C3
6115S1(I)=S
6120RETURN
630ADEFFN'23
G30UDEFFN'23
G302REW CALCULATES PENETRATION Q7=011
6305S=03*V0+C3
6310S=(B3*V+C3)/S:S=LOG(S)
63155=U3-57B3
6320S=(V0-V+S)/B3
6325S1(I)=S
6315S=C3*S/B3
6330RETURII
6600DEFF11'26
```

6602REM CALCULATES PENETRATION Q7=111.
6625Z1=A3\*V\*V+B3\*V+C3
6630Z1=(A3\*V0\*V0+B3\*V0+C3)/Z1
6635Z1=LOG(Z1)
6637S=B3\*T1(I)
6640S=0.5\*(Z1-S)/A3
6740 S1(I)=S
6745RETURN
7000DEFFN'01
7002REM PRINTS DASHED LINE
7005Z9\$="----"
7010PRINT Z9\$,Z9\$,Z9\$,Z9\$

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